Register Number

SATHYABAMA UNIVERSITY

(Established under section 3 of UGC Act,1956)

Course & Branch :B.E - AERO Title of the Paper :Heat Transfer Sub. Code :526603-26603 Date :09/11/2009

Max. Marks :80 Time : 3 Hours Session :AN

PART - A Answer ALL the Questions (10 x 2 = 20)

- 1. State the Fourier's law of heat conduction.
- 2. What are extended surfaces?
- 3. Explain the concept of free convection heat transfer mechanism.
- 4. Define Prandtl and Stanton numbers and give their importance in convection heat transfer.
- 5. Distinguish between black body and gray body.
- 6. What are radiation shields?
- 7. Give the classification of heat exchanger.
- 8. Define NTU in heat exchanger.
- 9. What is ablative heat transfer?
- 10. Define the thermal efficiency of a gas turbine.

PART – B Answer All the Questions

$(5 \times 12 = 60)$

- 11. (a) Derive an expression for one dimensional Radial heat conduction through hollow cylindrical system.
 - (b) A plane wall 10cm think generates heat at the rate of 4 x 10^4 W/m³ when an electric current is passed through it. The convective heat transfer co-efficient between each face of the wall and the ambient air is 50Wm²K. Determine.
 - (i) The surface temperature.
 - (ii) The maximum temperature in the wall.

Assume the ambient air temperature to be 20°C and the thermal conductivity of the wall material to be 15W/mK.

(or)

- 12. (a) Discuss heat transfer using lumped parameter analysis.
 - (b) A stainless ball diameter $2cm(d_o)(\rho = 7865kg/m^3, Cp = 0.46KJ/kg^{\circ}C and K = 61W/m^{\circ}C)$ is uniformly heated to a temperature $T_i = 800^{\circ}C$. It is to be hardened by suddenly dropping it onto an oil bath at $T_a = 50^{\circ}C$. If the quenching occurs when the ball reaches $100^{\circ}C(T)$ and the heat transfer co-efficient (h) for the oil and sphere is $300w/m^{2\circ}C$ (T) and the heat transfer co-efficient (h) for the oil and sphere is $300W/m^2$ °C, how long should the ball be kept in the oil bath?
- 13. (a) Sketch temperature and velocity profiles in free convection on a vertical wall.
 - (b) Air at atmospheric pressure is contained between two horizontal panels separated by a distance of 25.4mm. The temperature of the lower panel is 60°C and the upper panel is at 15.6°c. Calculate the free convection heat transfer per m^2 of the panel surface. At 37.8°C air will have f =

1.121kg/m³, K = 29.2 x 10⁻²W/mK, Pr = 0.7, $\gamma = 0.171$ x 10⁻⁴m²/s, $\beta = 3.22$ x 10⁻³K⁻¹. (or)

14. (a) In a long annulus (3.125 cm ID and 5cm O.D), the air is heated by maintaining the temperature of the outer surface of inner tube at 50°C. The air enters at 16°C and leaves at 32°C and its flow rate is 30m/s. Estimate the heat transfer co-efficient between the air and the inner tube. Air properties 24°C.

 $\rho = 1.614 kg / m^3$, $\gamma = 15.9 X 10^{-6} m^2 / s$, $C_p = 1.007 kJ / kg K$, Pr = 0.707, K = 0.023 W / mK.

- (a) Describe the relation between fluid friction and heat transfer. (4)
- 15. (a) Describe the radiation heat exchange between two large parallel gray planes.
 - (b) Emissiveties of two large parallel plates maintained at 800° C and 300° C are 0.3 and 0.5 respectively. Find the net radiant heat exchange per m² for these plates.

(or)

- 16. (a) State Stefan-Boltzmann law. (2)
 - (b) Explain Radiation shape factor. (4)
 - (c) Two circular discs of diameter 20cm each are placed 2m apart. Calculate the radiant heat exchange for these plates if these plates are maintained at 800°C and 300°C respectively and their corresponding emissivities are 0.3 and 0.5.
- 17. (a) Explain the concept of mean temperature difference.
 - (b) In a double pipe counter flow heat exchanger, 10,000kg/hr of an oil having a specific heat of 2095 J/kg°C is cooled from 80°C to 50°C by 8000kg/hr of water entering at 25°C. Determine the heat exchange area for an overall heat transfer co-efficient of $300W/m^2K$. Take C_p for water as 4180 J/kg K.

- 18. (a) Define the effectiveness of a heat exchanger.
 - (b) Explain the stepwise procedure to determine the area required for a heat exchanger by NTU method. (8)

(4)

19. Explain the heat transfer problems in gas turbine combustion chamber.

(or)

20. Explain the aerodynamic heating.