Seat No.:	Enrolment No.
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		B.E. Sem-V <sup>th</sup> Examination December 2010	
J		ode: 150104 Sub Name: Computational Fluid Dynamics-I 2 /2010 Time: 03.00 pm - 05.30 pm Total Marks: 70	
Instr			
	<b>2.</b> ]	Attempt all questions. Make suitable assumptions wherever necessary. Figures to the right indicate full marks.	
Q.1	(a)	Derive the energy equation for a viscous flow with heat transfer in non	07
	(b)	conservation form.  Justify CFD is a research tool. Write and explain the steps involved in CFD process.	07
Q.2	(a)	What is the difference between the Euler's model and Navier stokes model of	07
	(b)	equations? Write the generic form of Navier Stokes model. Explain the supersonic flow over the nose of a blunt body. Write a short note on shock capturing method.	07
	(b)	OR Consider the irrotational, 2-D, inviscid, steady flow of a compressible gas. The flow field is slightly perturbed from free seam like flow over a thin profile. Find the roots of equations involved in such kind of flow problem using Cramer's rule and Eigen method.	07
Q.3	(a)	Distinguish between the basic discretization techniques. Derive the expression for 1 <sup>st</sup> order forward, 1 <sup>st</sup> order rearward and 2 <sup>nd</sup> order central difference equation with	07
	(b)	respect to x.  Explain the domain and boundaries for the solution of parabolic equation in 2-D. also explain the steady boundary layer flows over a body.  OR	07
Q.3	(a)	Derive the differential equation to unsteady 1-D heat conduction equation. Also	07
	(b)	define the accurate solution and precise solution. Write a short note on Lax Wendroff method and gives its stability criteria.	07
Q.4	(a)	Differentiating between explicit approach and implicit approach for the solution of	07
	(b)	difference equations. Formulate the explicit form for 1-D heat conduction equation.  With an example explain the concept of compressed grid	07

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07

- **Q.4** Describe Mac Cormark multi-step method. (a)
  - Consider the viscous flow of air over a flat plate. Variation in velocity with respect 07 **(b)** to y is given as:  $u=1582(1-e^{-y/L})$ . Where L= 1 unit and  $\mu=3.37\times10^{-7}$  slug/ (ft.s). y is from 0 to 0.3 in the steps of 0.1. Find the percentage error in shear stress, involved in 1<sup>st</sup> ordered and 2<sup>nd</sup> ordered difference compared to exact solution.
- Draw and explain the subsonic-supersonic flow through the C-D nozzle and also **07 Q.5** show the variation in properties along the length of nozzle
  - Derive the momentum equation for the 2-D subsonic supersonic flow through C-D 07 nozzle.

OR

- How the boundary conditions and initial conditions are applied to the nozzle flow. **07 Q.5** (a)
  - Why the governing equations are to be transformed into non dimensional form? **07** Derive the momentum equation in non dimensional form for the nozzle flow.

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